

## *The Worst Response of Mistuned Bladed Disk System Using Genetic Algorithm*

Ehsan Raeisi and Saeed Ziaei-Rad

### **ABSTRACT**

The objective of this paper is to develop a technique using genetic algorithms (GA) for predicting the worst response of mistuned bladed disk. A simple but representative model for the bladed disk system is presented. The response of the system under harmonic excitation was obtained. The worst response of the mistuned system was then formulated as an optimization problem. Next, mistuning in blades stiffness, blades damping and blades stiffness and damping was studied and the results were compared. The results indicate that the system mistuned with both stiffness and damping has higher response amplitude in comparison with other mistuned system.

**KEYWORDS** : Bladed disk system, Mistuned System, Worst frequency response, Genetic algorithm

[ ]

---

/ / :

// :

ehsan1194@yahoo.com :

i

ii

szrad@cc.iut.ac.ir :

[ ]

[ ] [ ]  
%

[ ]

[ ] [ ]

( )

[ ]

$$\frac{1}{\sqrt{y}}(1 + \sqrt{N})$$

[ ]

N

[ ]

$$\frac{1}{\sqrt{y}}\left(1 + \sqrt{\frac{N}{y}}\right)$$

[ ]

[ ]

[ ]



[ ]

( )

( )

$$k = \text{N/m} \quad m = \text{kg} \quad c = \text{Ns/m} \quad (\%)$$

/ %

/ %

c

$\omega$

)

$(H(\omega))$

[ ]

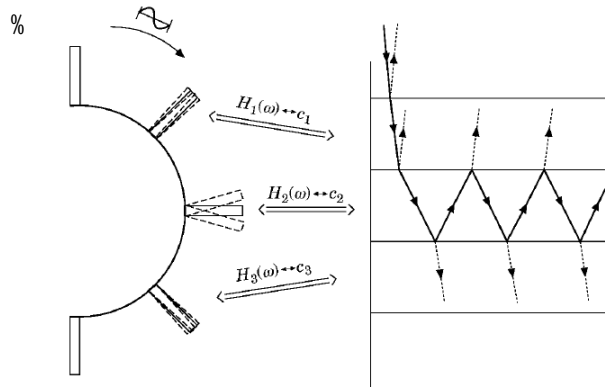
)

(

( [ ] [ ] )

/ %

( % )



( )

n

(

(

:( )

)

[ ]

:.....

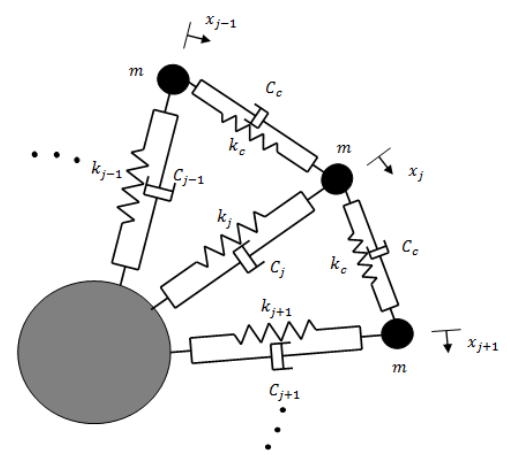
—



( )  
 $(c_c \quad k_c)$   
 $c_j$   
 $k_j$   
 $r$   
 $n$

( )  
 $[ \quad ] [ \quad ]$

( )  
 $M \ddot{\mathbf{X}} + C \dot{\mathbf{X}} + K \mathbf{X} = \mathbf{F}$   
 $n$   
 $K \quad C \quad M$   
 $\mathbf{X} ( )$   
 $n \times n$   
 $\mathbf{F}$



( )  
 $\mathbf{F} (t) = \mathbf{F}_0^{(c)} \cos \omega t + \mathbf{F}_0^{(s)} \sin \omega t$   
 $( )$   
 $( )$   
 $\mathbf{X} (t) = \mathbf{U} \cos \omega t + \mathbf{V} \sin \omega t$   
 $( )$   
 $( ) ( )$

$\mathbf{V} \quad \mathbf{U}$   
 $( )$   
 $H \mathbf{Z} = \tilde{\mathbf{F}}$   
 $( )$

$\mathbf{Z}^T = [U_1 \quad V_1 \quad U_2 \quad V_2 \quad \dots \quad U_n \quad V_n]$   
 $( )$

$\tilde{\mathbf{F}}^T = [F_{01}^{(c)} \quad F_{01}^{(s)} \quad F_{02}^{(c)} \quad F_{02}^{(s)} \quad \dots \quad F_{0n}^{(c)} \quad F_{0n}^{(s)}]$   
 $( )$

$\mathbf{H}$   
 $n \times n$   
 $( ) ( )$

$\omega C_{il} \quad K_{il} - \omega^2 M_{il}$   
 $(i, l = 1, \dots, n)$   
 $\mathbf{Z}$   
 $i$



$r$   $\omega_r$

$$[428600, 431400] N/m$$

$(k_t)$  /  $( )$

$$[.129, .149] Ns/m$$

$n$   
 $n$

$$c_c = 1/443 Ns/m$$

[ ] [ ]

( )

:( )

|   |  |  |
|---|--|--|
| % |  |  |
| % |  |  |
|   |  |  |

)

(

) ( )

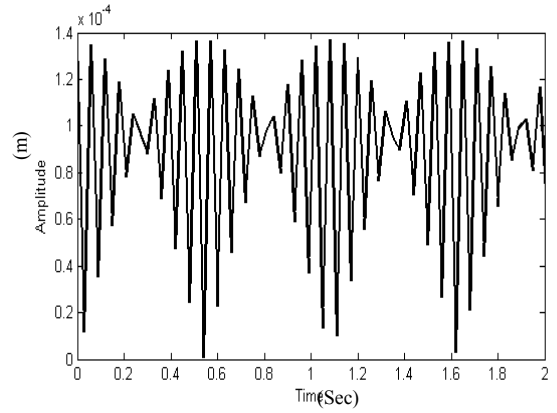
.(

$\phi_i$

( )

( )

$$1/2680 \times 10^{-4}$$



:( )

$k_t =$  N/m  $k_c =$  N/m  $m =$  / kg  
 $c_r =$  / Ns/m  $c_c =$  / Ns/m  
 $r$

$$F_i(t) = F_0 \cos(\omega_r t - \phi_i) \quad ( )$$

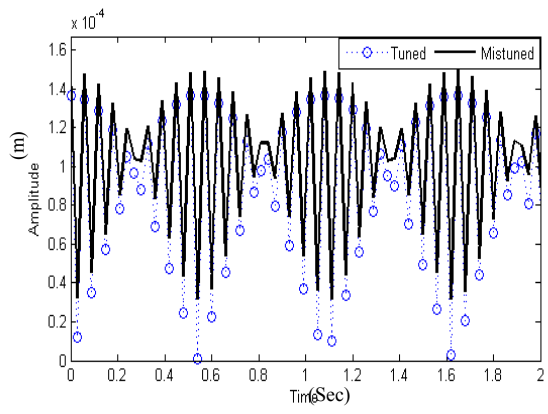
:[ ] ( )

$$\omega_r^2 = \frac{k_t}{m} + 4 \frac{k_c}{m} \sin^2 \frac{\pi r}{N} \quad ( )$$

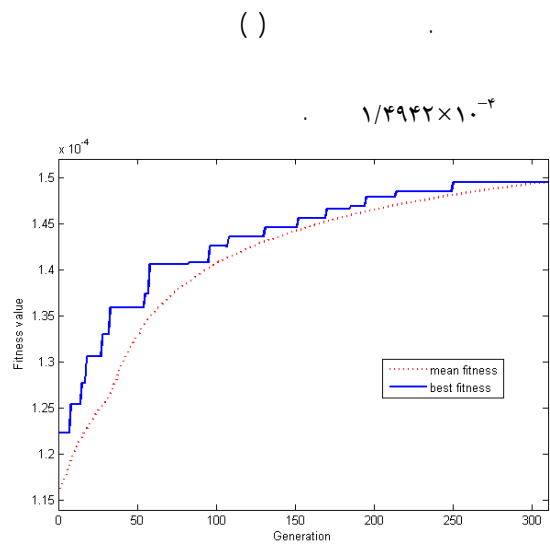
$$\phi_i = \frac{2\pi}{N} r(i-1) \quad ( )$$



/ / / /



( )



( )

( )

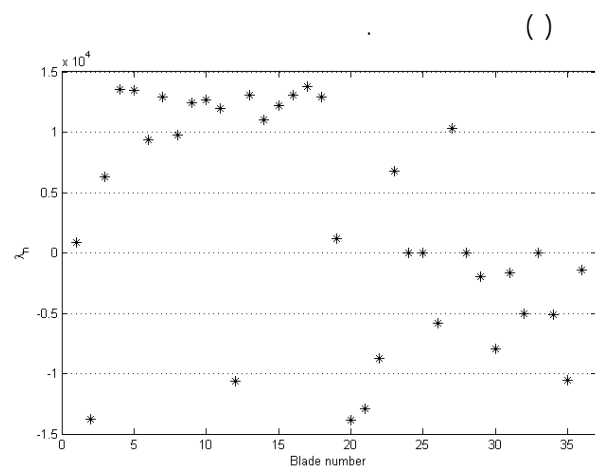
( ) ( )

( ) /  $1/2680 \times 10^{-4}$

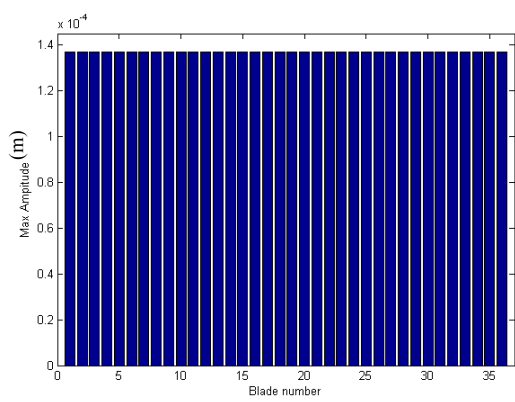
( )  $1/4942 \times 10^{-4}$

%

%



( )

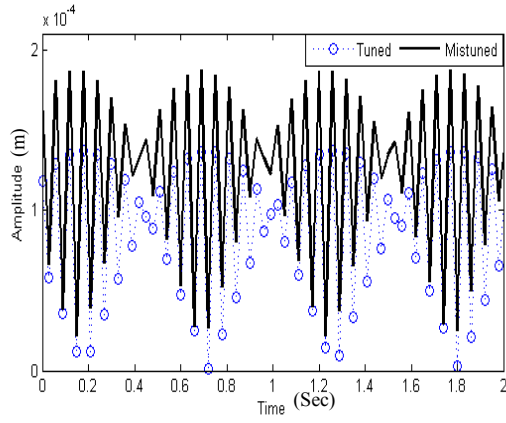


( )

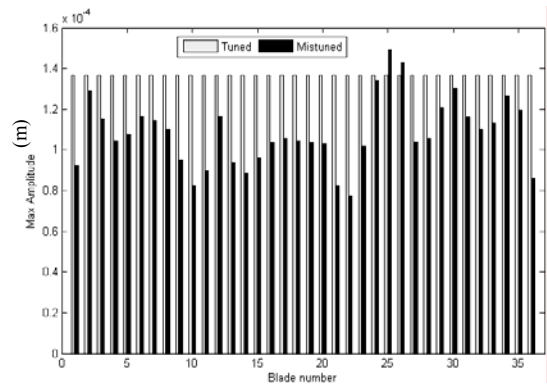
( )

)

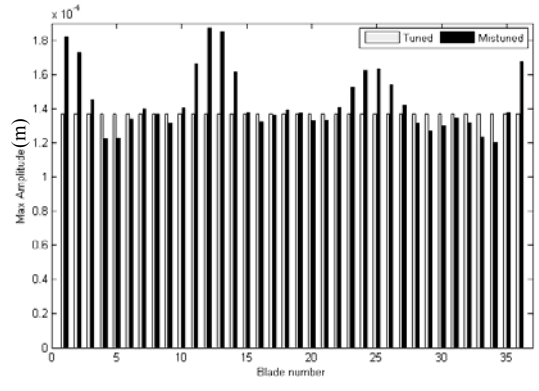
(



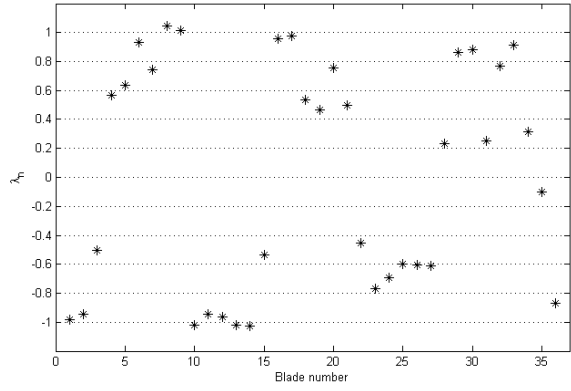
( )



( )



( )



( )

( )

( )

( )

( )

( )

$$1/18112 \times 10^{-4}$$

( )

( )

( )

( )

( )

$$1/26880 \times 10^{-4}$$

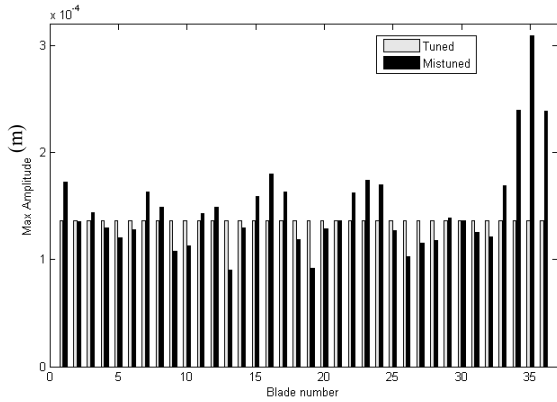
$$( 1/18112 \times 10^{-4}$$





$$1/2685 \times 10^{-4}$$

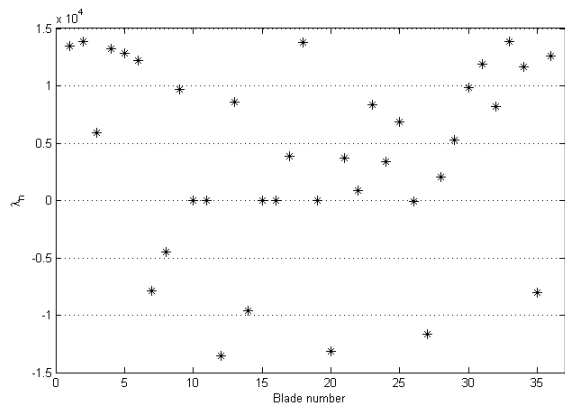
$$3/0.926 \times 10^{-4}$$



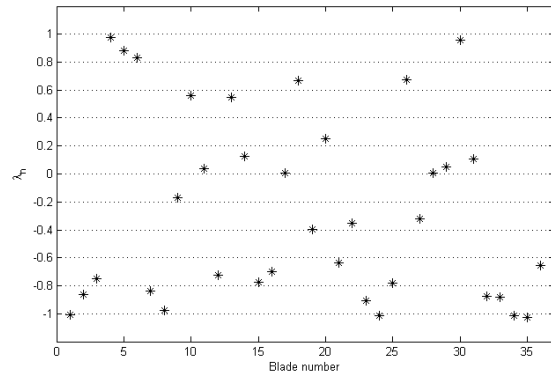
( )

( )

( )



( )



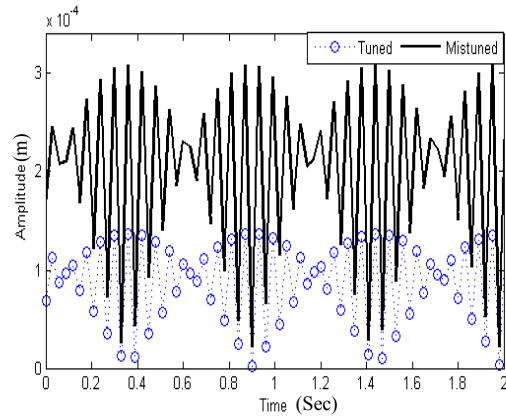
( )

( )

)

(

$$3/0.926 \times 10^{-4}$$



( )

( )

( )

%

/ %



:( )

|   |                          |  |  |
|---|--------------------------|--|--|
|   |                          |  |  |
|   |                          |  |  |
|   |                          |  |  |
| / | $7/92.6 \times 10^{-5}$  |  |  |
| / | $7/98.1 \times 10^{-5}$  |  |  |
| / | $8/2285 \times 10^{-5}$  |  |  |
| / | $9/2666 \times 10^{-5}$  |  |  |
|   |                          |  |  |
| / | $1/0.642 \times 10^{-4}$ |  |  |
| / | $1/1789 \times 10^{-4}$  |  |  |
| / | $1/20.21 \times 10^{-4}$ |  |  |
| / | $1/2521 \times 10^{-4}$  |  |  |
|   |                          |  |  |
| / | $1/2523 \times 10^{-4}$  |  |  |
| / | $1/4.51 \times 10^{-4}$  |  |  |
| / | $1/4422 \times 10^{-4}$  |  |  |
| / | $2/15.9 \times 10^{-4}$  |  |  |
|   |                          |  |  |
| / | $1/3685 \times 10^{-4}$  |  |  |
| / | $1/4942 \times 10^{-4}$  |  |  |
| / | $1/8712 \times 10^{-4}$  |  |  |
| / | $2/0.926 \times 10^{-4}$ |  |  |

- Afolabi, D.H., *Vibration of Mistuned Bladed Disc Assembly*, PhD Thesis, Imperial College of Science, Technology & Medicines, University of London, 1982. [ ]
- Afolabi, D.H. "Vibration Amplitudes of Mistuned Blades", *ASME, Journal of Turbomachinery*, Vol. 110, pp. 251–257, 1988. [ ]
- Lin C.C., Mignolet M.P., "Effects of Damping and Damping Mistuning on the Forced Vibration Response of Bladed Disks", *J. Sound Vib.* Vol. 193, pp. 525-543, 1996. [ ]
- Y.J. Chiu, Sh.Ch. Huang, "The influence on coupling vibration of a rotor system due to a mistuned blade length", *Journal of Mechanical Sciences*, Vol. 49, pp. 522–532, 2007. [ ]
- Y.J. Yan, P.L. Cui, H.N. Hao, "Vibration mechanism of a mistuned bladed-disk", *Journal of Sound and Vibration*, Vol. 317, pp. 294–307, 2008. [ ]
- Y.-J. Chan, D.J. Ewins, "Management of the variability of vibration response levels in mistuned bladed discs using robust design concepts. Part 1 Parameter design", *J. Mechanical Systems and Signal Processing*, Vol. 24, pp. 2777–2791, 2010. [ ]
- J.H. Griffin, A. Sinha, "The interaction between mistuning and friction in the forced response of bladed disk assemblies", *Journal of Engineering for Gas Turbines and Power*, Vol. 107, pp. 205–211, 1985. [ ]
- J.H. Griffin, A. Sinha, "The interaction between mistuning and friction in the forced response of bladed disk assemblies", *Journal of Engineering for Gas Turbines and Power*, Vol. 107, pp. 205–211, 1985. [ ]
- D.S. Whitehead, "Effects of mistuning on the vibration of turbomachinery blades induced by wakes", *J. Mechanical Engineering Science*, Vol.8, No. 1, pp. 15–21, 1966. [ ]
- D.S. Whitehead, "Effects of Mistuning on Forced Vibration of Blades with Mechanical Coupling", *Journal of Mechanical Engineering Science*, Vol.18, No. 6, pp. 306-307, 1976. [ ]
- R.C.F. Dye, T.A. Henry, "Vibration amplitudes of compressor blades resulting from scatter in blade natural frequencies", *ASME, Journal of Engineering for Power*, Vol. 91, No. 3, pp. 182-188, 1969. [ ]
- Wagner, J.T., "Coupling of turbomachine blade vibrations through the rotor", *ASME, Journal of Engineering for Power*, Vol. 89, No. 4, pp. 502–512, 1967. [ ]
- D.J. Ewins, "The effect of detuning upon the forced vibration of bladed disks", *J. Sound and Vibration*, Vol.9, pp. 65–79, 1969. [ ]
- L.E. EL-Bayoumy, A.E. Srinivasan, "Influence of mistuning on rotor blade vibration", *AIAA Journal*, Vol. 13, No. 4, pp. 460–464, 1975. [ ]



- M.P. Mignolet, C.C. Lin, "The combined closed form-perturbation approach to the analysis of mistuned bladed disks", *Journal of Turbomachinery*, Vol. 115, pp. 771–780, 1993. [ ]
- C.C. Lin, The combined closed form-perturbation approach to the analysis of mistuned bladed disks, M.S. Thesis, Arizona State University, 1991. [ ]
- Goldberg, D. E., *Genetic Algorithms in Search, Optimization, and Machine Learning*, Addison-Wesley, Reading, MA, 1989. [ ]
- Davis (Ed.), L., *Handbook of Genetic Algorithms*, Van Nostrand Reinhold, 1991. [ ]
- J.H. Griffin, "On predicting the resonant response of bladed disk assemblies", *Journal of Engineering for Gas Turbines and Power*, Vol. 110, pp. 45–50, 1988. [ ]
- R.L. Jay, D.W. Burns, "Characteristics of the diametral resonant response of a shrouded fan under a prescribed distortion", *Journal of Vibration, Acoustics, Stress and Reliability in Design*, Vol. 108, pp. 125–131, 1986. [ ]
- A. Sinha, S. Chen, "A higher order technique to compute the statistics of forced response of a mistuned bladed disk", *Journal of Sound and Vibration*, Vol. 130, pp. 207–221, 1989. [ ]
- A. Sinha, "Calculating the statistics of forced response of a mistuned bladed disk assembly", *American Institute of Aeronautics and Astronautics Journal*, Vol. 24, pp. 1797–1801, 1986. [ ]

- 
- <sup>1</sup> Whitehead
  - <sup>2</sup> Rolls Royce
  - <sup>3</sup> NASA
  - <sup>4</sup> Dye & Henry
  - <sup>5</sup> Single Mistuning
  - <sup>6</sup> Alternative Mistuning
  - <sup>7</sup> Random Mistuning
  - <sup>8</sup> Wagner
  - <sup>9</sup> Ewins
  - <sup>10</sup> Han
  - <sup>11</sup> Afolabi
  - <sup>12</sup> Lin
  - <sup>13</sup> Mignolet
  - <sup>14</sup> Chiu
  - <sup>15</sup> Huang
  - <sup>16</sup> Yan
  - <sup>17</sup> Chun
  - <sup>18</sup> Multipoint
  - <sup>19</sup> Multi-nonuniform