

مهدی قاسمیه*^۱ و مهدی کریمی راد^۲

(// // //)

()

[] [] t n $[0, t]$ $(t > 0)$

Δt $\Delta t = t/n$ Δt

[] []

[]

[]

[]

() ()

() () $\alpha \delta$
 $t + \Delta t$

${}^{t+\Delta t}\ddot{U}$ () ()
 $t + \Delta t$

$t + \Delta t$
 $t - \Delta t \quad t$

${}^0\dot{U}$ 0U $M\ddot{U} + C\dot{U} + KU = P$ (۱)
 $()$ $K \quad C \quad M$
 P

${}^0\ddot{U}$ $t = 0$
 $() ()$ $\ddot{U} \quad \dot{U} \quad U$
 Δt t

() () ${}^t\ddot{U} \quad {}^t\dot{U} \quad {}^tU$
 $n\Delta t \quad \dots \quad 3\Delta t \quad 2\Delta t$ t ${}^{t+\Delta t}\dot{U}$ ${}^{t+\Delta t}U$ () ()

$\delta \quad \alpha$ () () () ()
 $t - \Delta t$ $t + \Delta t$

() () () ()

${}^{t+\Delta t}U = {}^tU + \Delta t {}^t\dot{U} + \frac{\Delta t^2}{2} {}^t\ddot{U} + \frac{\Delta t^3}{6} {}^t\dddot{U} + \dots$ (۲)

${}^{t+\Delta t}\dot{U} = {}^t\dot{U} + \Delta t {}^t\ddot{U} + \frac{\Delta t^2}{2} {}^t\dddot{U} + \frac{\Delta t^3}{6} {}^t\ddot{\ddot{U}} + \dots$ (۳)

${}^{t+\Delta t}U = {}^tU + \Delta t {}^t\dot{U} + \frac{\Delta t^2}{2} {}^t\ddot{U} + \frac{\Delta t^3}{6} {}^t\dddot{U} + \alpha \Delta t^4 {}^t\ddot{\ddot{U}}$ (۴)

${}^{t+\Delta t}\dot{U} = {}^t\dot{U} + \Delta t {}^t\ddot{U} + \frac{\Delta t^2}{2} {}^t\dddot{U} + \delta \Delta t^3 {}^t\ddot{\ddot{U}}$ (۵)

${}^t\ddot{\ddot{U}} = \frac{1}{2\Delta t} ({}^{t+\Delta t}\ddot{U} - {}^{t-\Delta t}\ddot{U})$ (۶)

${}^t\ddot{\ddot{U}} = \frac{1}{\Delta t^2} ({}^{t-\Delta t}\ddot{U} - 2{}^t\ddot{U} + {}^{t+\Delta t}\ddot{U})$ (۷)

$${}^{t+\Delta t}\dot{U} = {}^t\dot{U} + \left[\left(\delta - \frac{1}{4} \right) {}^{t-\Delta t}\ddot{U} + (1-2\delta) {}^t\ddot{U} + \left(\delta + \frac{1}{4} \right) {}^{t+\Delta t}\ddot{U} \right] \Delta t \quad (۸)$$

$${}^{t+\Delta t}U = {}^tU + {}^t\dot{U} \Delta t + \left[\left(\alpha - \frac{1}{12} \right) {}^{t-\Delta t}\ddot{U} + \left(\frac{1}{2} - 2\alpha \right) {}^t\ddot{U} + \left(\alpha + \frac{1}{12} \right) {}^{t+\Delta t}\ddot{U} \right] \Delta t^2 \quad (۹)$$

$$T = 2\pi / \omega$$

() () []

x

() ()
() () ()

p k c m
t + Δt

$${}^{t+\Delta t}\ddot{x} \quad {}^{t+\Delta t}\ddot{x}$$

() ()

$${}^{t+\Delta t}\ddot{x} + 2\xi\omega {}^{t+\Delta t}\dot{x} + \omega^2 {}^{t+\Delta t}x = {}^{t+\Delta t}r \quad (۱۰)$$

: ()

$$\omega^2 = k/m \quad 2\xi\omega = c/m$$

$$\omega \quad {}^{t+\Delta t}r = {}^{t+\Delta t}p/m$$

$${}^{t+\Delta t}\dot{x} = {}^t\dot{x} + \left[\left(\delta - \frac{1}{4} \right) {}^{t-\Delta t}\ddot{x} + (1-2\delta) {}^t\ddot{x} + \left(\delta + \frac{1}{4} \right) {}^{t+\Delta t}\ddot{x} \right] \Delta t \quad (۱۱)$$

$${}^{t+\Delta t}x = {}^tx + {}^t\dot{x} \Delta t + \left[\left(\alpha - \frac{1}{12} \right) {}^{t-\Delta t}\ddot{x} + \left(\frac{1}{2} - 2\alpha \right) {}^t\ddot{x} + \left(\alpha + \frac{1}{12} \right) {}^{t+\Delta t}\ddot{x} \right] \Delta t^2 \quad (۱۲)$$

$$\begin{Bmatrix} {}^{t+\Delta t}\ddot{x} \\ {}^t\ddot{x} \\ {}^{t+\Delta t}\dot{x} \\ {}^{t+\Delta t}x \end{Bmatrix} = [A] \begin{Bmatrix} {}^t\ddot{x} \\ {}^{t-\Delta t}\ddot{x} \\ {}^t\dot{x} \\ {}^tx \end{Bmatrix} + \{L\} {}^{t+\Delta t}r \quad (۱۳)$$

$$\lambda^3 + \frac{(-2 + h(\delta - 3\alpha + \frac{2}{3}))}{(1 + h(\alpha + \frac{1}{12}))} \lambda^2 + \frac{(1 + h(3\alpha - 2\delta + \frac{5}{12}))}{(1 + h(\alpha + \frac{1}{12}))} \lambda + \frac{h(\delta - \alpha - \frac{1}{6})}{(1 + h(\alpha + \frac{1}{12}))} = 0 \quad (۱۴)$$

$$\rho(A) \leq 1$$

$$\{L\} [A]$$

[]

[]

{L}

[A]

$$\rho(A) [A]$$

α δ

:

[A]

$$\rho(A) = \max_i |\lambda_i| \quad (۱۵)$$

[A]

()

[A]

λ_i ()

α δ

()

λ_i

$$|\lambda_i| = (\lambda_i \cdot \bar{\lambda}_i)^{1/2}$$

λ_i

$\delta = 1/3$

()

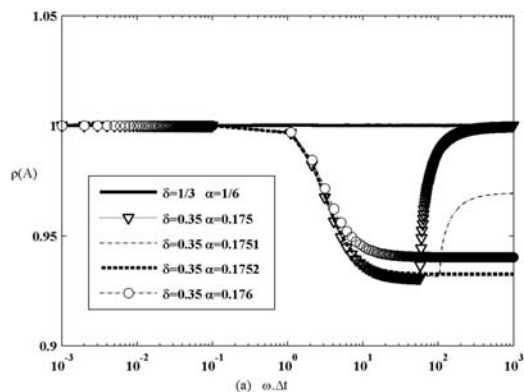
$\rho(A) \leq 1$

[A]

$\alpha \quad \delta$

$\alpha = 1/6$

$\delta \geq 1/3; \delta/2 \leq \alpha \leq \delta - 1/6$ (16)



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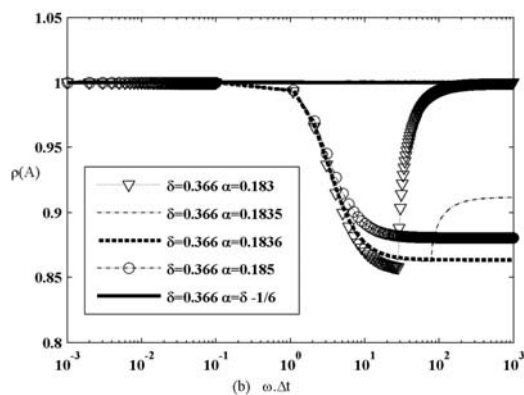
$\alpha = 1/6 \quad \delta = 1/3$

(a)

$\delta = 0.4 \quad \delta = 0.366 \quad \alpha = \delta - 1/6$

[A]

(c) (b)



$(\rho_\infty = \lim_{\omega \cdot \Delta t \rightarrow \infty} \rho(A)) \rho_\infty$

[]

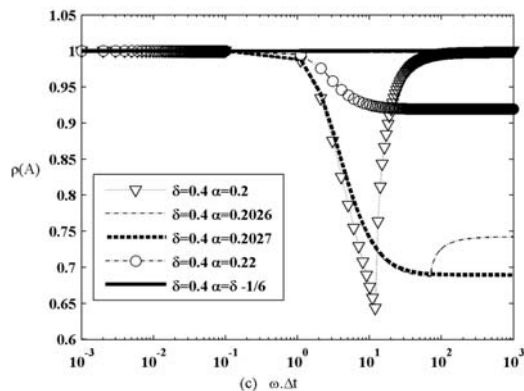
$\alpha = 1/6 \quad \delta = 1/3$

α

$\delta > 1/3$

ρ_∞

(a)



شکل ۱ : منحنی شعاع طیفی ماتریس [A] بر حسب $\omega \cdot \Delta t$.

()

$\delta = 0.4 \quad \delta = 0.366$

(c) (b)

$\alpha = 0.1836$

ρ_∞

$\alpha = 0.2027$

()

[]

$$\alpha \quad \delta$$

()

[A]

$$\alpha = 1/6 \quad \delta = 1/3$$

()

δ

α

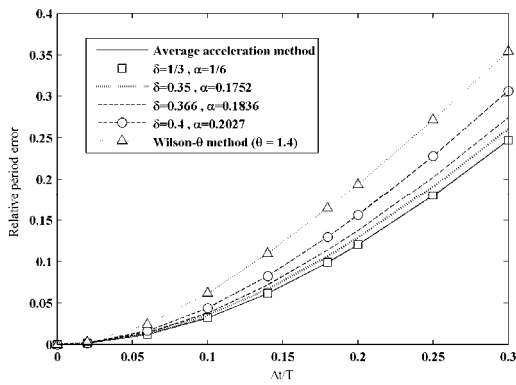
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$$\delta = 0.366)$$

$$(\alpha = 0.2027 \quad \delta = 0.4) \quad (\alpha = 0.1836$$

α

$\delta > 1/3$



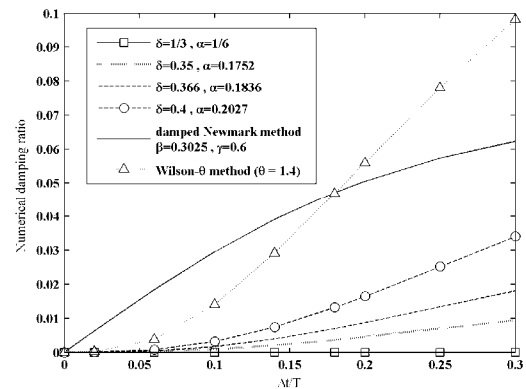
[]

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$$\Delta t/T \quad \alpha = 0.1836 \quad \delta = 0.366$$

$$\alpha = 0.2027 \quad \delta = 0.4 \quad 0.115$$

شکل ۲ : مقایسه نسبت میرایی عددی روش پیشنهادی، روش ویلسون و روش نیومارک.

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$\bar{\xi}$

[]

$$AD = 2\pi\bar{\xi}$$

$$T = 2\pi \quad x_{exact} = \cos(t) \quad 0.107 \quad \Delta t/T$$

$$(\quad) \quad (\quad) \quad (\quad) \quad (\quad)$$

$$(\quad) \quad \Delta t = T/10 \quad [A]$$

$$\alpha = -0.3$$

$$[\quad] \quad \theta = 1.4$$

$$0.04 \quad 0.08 \quad 0.1 \quad \Delta t/T$$

$$[\quad]$$

$${}^t e = \frac{|{}^t x - {}^t x_{exact}|}{{}^t x_{exact}} \quad (18) \quad \alpha = 0.1836 \quad \delta = 0.366$$

$${}^t x \quad {}^t x_{exact}$$

$$t$$

$$(\quad)$$

$$\ddot{x} + x = 0 \quad (19)$$

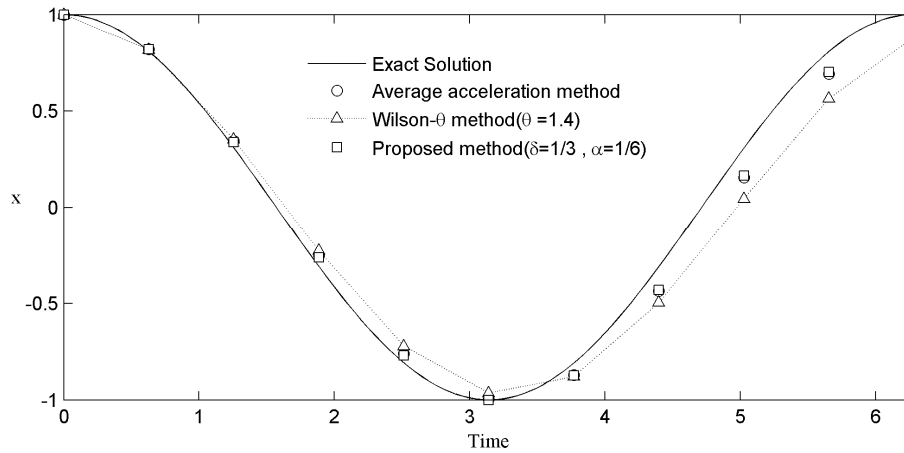
$$(\quad) \quad \dot{x} = 0 \quad x = 1$$

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جدول ۱: حل عددی معادله دیفرانسیل (۱۷) با استفاده از روش‌های شتاب متوسط، ویلسون و پیشنهادی.

Time			$(\theta = 1.4)$		$(\delta = 1/3, \alpha = 1/6)$	
	${}^t x$	${}^t e$	${}^t x$	${}^t e$	${}^t x$	${}^t e$
Δt	0.8203	0.013968	0.8187	0.01199	0.8203	0.013968
$2\Delta t$	0.3459	0.119411	0.3529	0.142063	0.3405	0.101936
$3\Delta t$	-0.2528	-0.18187	-0.2273	-0.26439	-0.2616	-0.15339
$4\Delta t$	-0.7607	-0.0597	-0.7220	-0.10754	-0.7698	-0.04845
$5\Delta t$	-0.9952	-0.0048	-0.9651	-0.0349	-1.0013	-0.0013
$6\Delta t$	-0.8722	-0.07812	-0.8785	-0.08578	-0.8731	-0.07923
$7\Delta t$	-0.4357	-0.41001	-0.4968	-0.60741	-0.4311	-0.39512
$8\Delta t$	0.1573	0.490912	0.0464	0.850115	0.1658	0.463729
$9\Delta t$	0.6938	0.142395	0.5649	0.301724	0.7031	0.1309
$10\Delta t$	0.9810	0.019	0.8843	0.1157	0.9878	0.0122



شکل ۴: منحنی پاسخ معادله دیفرانسیل (۱۷) با استفاده از حل دقیق و روش‌های شتاب متوسط، ویلسون و پیشنهادی.

- 1 - Belytschko, T. and Lu, Y. (1993). "Explicit multi-time step integration for first and second order finite element semi-discretizations." *Computer Methods in Applied Mechanics and Engineering*, Vol. 108, PP. 353–383.

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- 2 - Hahn, G. D. (1991). "A modified Euler method for dynamic analysis." *International Journal for Numerical Methods in Engineering.*, Vol. 32, No. 5, PP. 943–955.
 - 3 - Hughes, T.J.R., and Belytschko, T. (1983). "A précis of developments in computational methods for transient analysis." *Journal of Applied Mechanics.*, Vol. 50, PP. 1033–1041.
 - 4 - Dokainish, M.A. and Subbaraj, K. (1989). "A survey of direct time integration methods in computational structural dynamics-I. explicit methods." *Computers & Structures*, Vol. 32, No. 6, PP. 1371-1386.
 - 5 - Hughes, T.J.R. (1987). *The finite element method: linear static and dynamic finite element analysis*. Prentice Hall, Englewood Cliffs, N.J.
 - 6 - Newmark, N.M. (1959). "A method of computation for structural dynamics." *Journal of the Engineering Mechanics Division., ASCE*, Vol. 85, No. 3, PP. 67–94.
 - 7 - Wilson, E.L., Farhoomand, I., and Bathe, K.J. (1973). "Nonlinear dynamic analysis of complex structures." *International Journal of Earthquake Engineering and Structural Dynamics.*, Vol. 1, No. 3, PP. 241–252.
 - 8 - Bathe, K.J., Wilson, E.L. (1973). "Stability and accuracy analysis of direct integration methods." *International Journal of Earthquake Engineering and Structural Dynamics.*, Vol. 1, No. 3, PP. 283–291.
 - 9 - Keierleber, C.W., Rosson, B.T. (2005). "Higher-order implicit dynamic time integration method." *Journal of Structural Engineering., ASCE*, Vol. 131, No. 8, PP. 1267–1276.
 - 10 - Bathe, K.J. (1996). *Finite element procedures*. Prentice-Hall, Englewood Cliffs, N.J.
 - 11 - Hilber, H.M., Hughes, T.J.R., and Taylor, R.L. (1977). "Improved numerical dissipation for time integration algorithms in structural dynamics." *International Journal of Earthquake Engineering and Structural Dynamics.*, Vol. 5, PP. 283–292.
 - 12 - Houbolt, J.C. (1950). "A recurrence matrix solution for the dynamic response of elastic aircraft." *Journal of the Aeronautical Sciences.*, Vol. 17, PP. 540–550.
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